BEST COPY Available

25 YEAR RE-REVIEW

Approved For Release 2009/07/07 : CIA-RDP67B00657R000100060001-3

SECURA NO. 86. 17204-63 COPY: 3

SOARD RADAR - AN/APQ-93

Review of Program since July 19, 1962

November 30, 1962

Ву

Westinghouse Electric Corporation

Air Arm Division

Baltimore 3, Maryland

This document contains information affecting the National Defense of the United States within the meaning of Espionage Laws, Title 18, U.S.C., Sections 793 and 794. Its transmission or the revelation of its contents in any manner to an unauthorized person is prohibited by law.

DOWNGRADED AT 12 YÉAR INTER-VALS: NOT AUTOMATICALLY DECLASSIFIED. DOD DIR 5200.10

IF ENCLOSURES ARE WITHDRAWN (OR NOT ATTACHED), THE CLASSIFICATION OF THIS CORRESPONDENCE WILL BE CELLED IN ACCORDANCE WITH PAR 25E AF REGULATION 2 OR NAVY REGULATION ARTICLE 76 (5) (H).

SECRET

INDEX

Page No.	<u>Title</u>
1	Cover
2	Index
3	Program for SOARD AN/APQ-93
4	Resonant Ring Improvement
5	Antenna Improvement
6	CFA Transmitter Summary
7	Breadboard Transmitter, Intermediate Amplifier
8	Breadboard Transmitter, CFA Modulator, Front View
9	Breadboard Transmitter, CFA Modulator, Rear View
10	Cross-Field Amplifier Waveforms
11	Elements of Motion Compensation
12	Design Evaluation
13	Environmental Test
זוי	Flight Test
15	Correlated Map Test Flight S-11
16	Correlated Map Test Flight S-33
17	Aerial Photo - South River
18 .	Target Description
19	Target Description
20	Correlated Data
21	Accessory Material and Service
22	Installation
23	Radar Block Diagram AN/APQ-93
24	Transmitter Block Diagram
25	Radar Parameters
26	System Weight

PROGRAM FOR SOARD AN/APQ-93

L CONTINUED DEVELOPMENT

A. RESONANT RING IMPROVEMENT

B. ANTENNA IMPROVEMENT

C. RECORDER SPOT SIZE REDUCTION STUDY

2. RADAR MODIFICATION

A. TRANSMITTER, CROSSED-FIELD AMPLIFIER B. IMPROVED MOTION COMPENSATION SYSTEM

C. RECORDER

LENS OPTICS REPLACING FIBER OPTICS AUTOMATIC FILM SPEED CONTROL

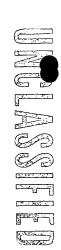


4 ENVIRONMENTAL TEST

5. FLIGHT TEST

6. ACCESSORIES

7. INSTALLATION





RESONANT RING IMPROVEMENT

! ORIGINAL UNIT		GOAL 0.5-1.0 MEG W.	MEASURED 0.23 MEG W
		10 NANOSEC. 20-40 W. AVGE.	10 NANOSEC. 9.2 W. AVGE.
2 Limiting factors	DRIVING POWER LOSSES IN RING		
3. IMPROVEMENTS	RING LENGTH INCREASED		0.14 MEG W.
	Tuning shorts improved		20 NANOSEC. 12.3 W. AVGE.
4 FURTHER IMPROVEMENTS	INCREASE RING LENGTH (FOLD)	0.40 MEG W.	
		30 NANOSEC.	

48 W. AVGE.

INCREASE DRIVE POWER



ANTENNA IMPROVEMENT

I. SIGNIFICANT FACTORS - HIGH TEMP. DESIGN 550°F -GAIN A SPECIAL TECHNIQUES FOR HIGH TEMP -STAINLESS STEEL BEAM

-ELECTRO-DEPOSITED NICKEL ELEMENTS -FIBRE GLASS-RESIN PRESSURE COVERS

-HIGH TEMP CEMENT

B. RESULTING UNIT HAD EXCESSIVE LOSSES

MAX. THEO. GAIN CALCUL, LOSSES PREDICTED GAIN

MEASURED GAIN

.5db

MANIFOLD PHASE CORRECTION

C IMPROVEMENTS

HIGH PRESSURE ADHESION

PREDICTED IMPROVED GAIN

CFA TRANSMITTER SUMMARY

I. SIGNIFICANT FACTORS

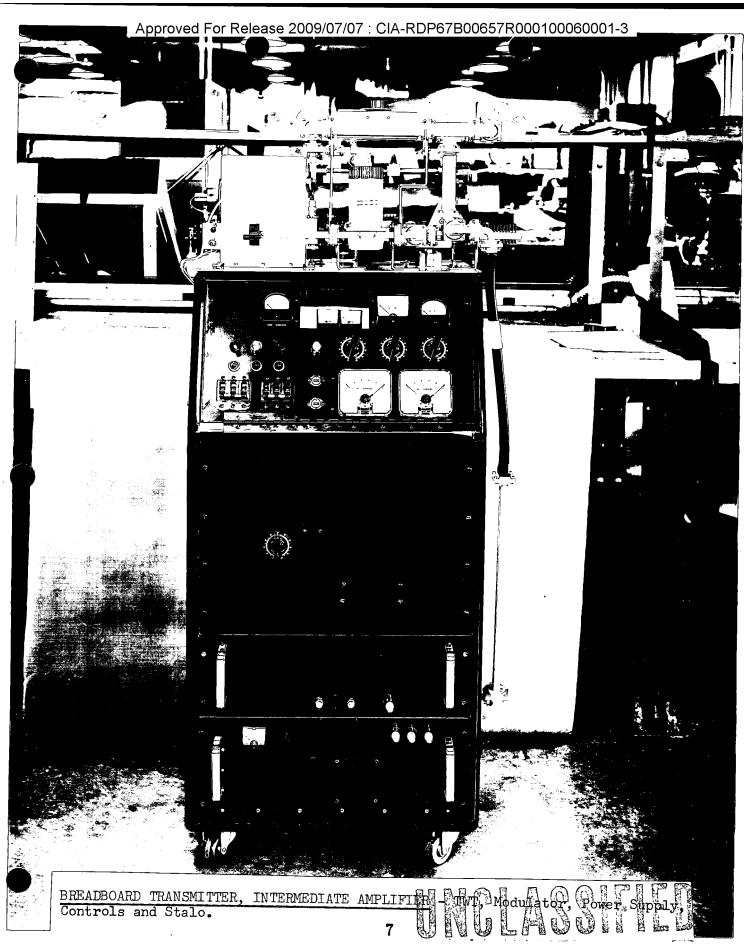
- A. POWER OUTPUT.......1-2 MEGAWATT
- PULSE TO PULSE PHASE STABILITY < 5° b. STABILITY......
- c. PULSE WIDTH.....30 NANOSEC.

2. SPECIAL TECHNIQUES

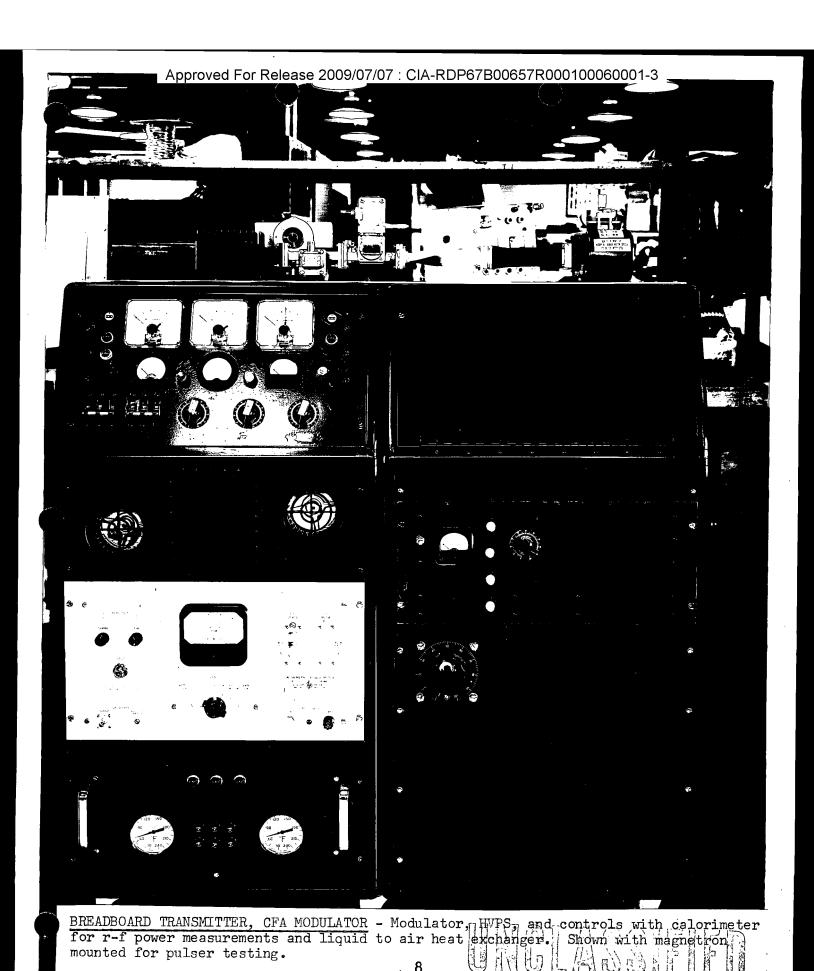
- a. CROSS-FIELD AMPLIFIER DEVELOPMENT
- b. MATCH CHARACTERISTICS OF POWER SUPPLIES, MODULATORS, AND RF AMPLIFIERS
- DARLINGTON LINE MODULATOR FOR CFA AND GRID PULSED TWT FOR INTERMEDIATE AMPLIFIER رن ن
- DIELECTRIC STRENGTH AND IMPROVED COMPONENT COOLING EFFICIENCY MINIMUM WEIGHT DESIGN BY USING SF& INSTEAD OF AIR FOR HIGHER -0i

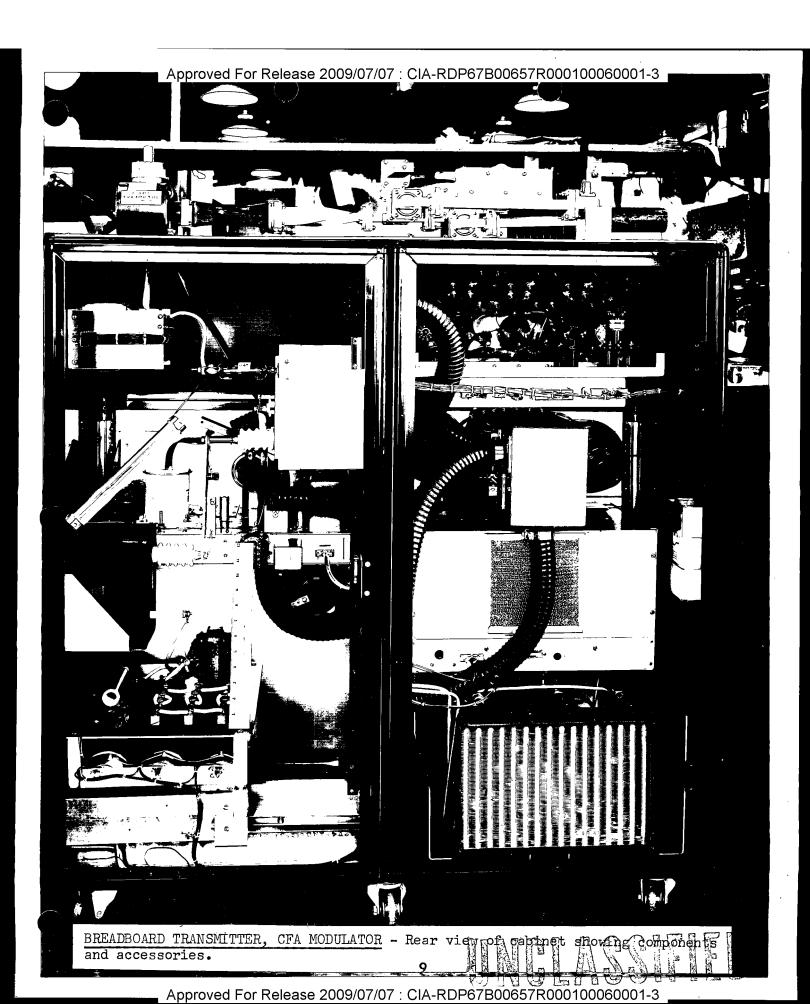
3 PROGRAM

- a. BUILT AND TESTED LABORATORY BREADBOARD MODULATOR
- b. SFD FOR CFA TUBE DEVELOPMENT (1st TUBE DEL. 12-17-62)
- c. FLYABLE BREADBOARD 1-15-63
- (TO BE DESIGN 2 PROTOTYPES DELIVERED 6-1-63 AND 7-1-63 IMPROVEMENTS OF FLYABLE BREADBOARD) Ö



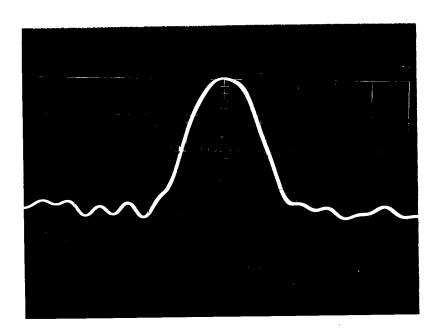
Approved For Release 2009/07/07 : CIA-RDP67B00657R000100060001-3



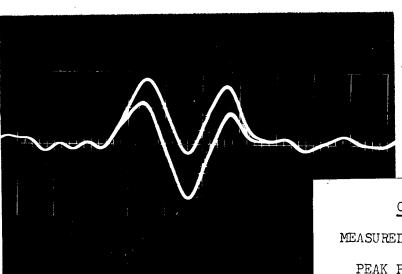


Approved For Release 2009/07/07: CIA-RDP67B00657R000100060001-3

CROSS-FIELD AMPLIFIER WAVEFORMS



AMPLITUDE DETECTED R-F PULSE



PHASE DETECTED R-F PULSE (REFERENCE PHASE-SHIFTED BY 20° TO SHOW PHASE JITTER SENSITIVITY OF 15°/CM)

CROSS-FIELD AMPLIFIER STATUS

MEASURED RESULTS:

PEAK POWER:

600 KW

PULSE WIDTH (-3 db)

40 NANOSEC 1.5° PK-PK +30°

PULSE-TO-PULSE PHASE DEVIATION: INTRAPULSE PHASE VARIATION:

GAIN (OVERDRIVEN CONDITION):

16 db

(BOTH PICTURES ARE 1 SECOND EXPOSURE, COMPARABLE TO RADAR DWELL TIME)

CONTROLS VARIABLE FREQUENCY GENERATOR TRANSVERSE VELOCITY CORRECTION ACCEL'EROMETER-ROLL STABILIZED INTEGRATION TO GIVE VELOCITY **ACTUATOR** 9 ANTENNA ELEMENTS OF MOTION COMPENSATION INTEGRATOR TRANSMITTER RECEIVER CORRECTS FOR PITCH AND YAW ERRORS YAW DRIVE WAR.FREQ.GEN ANTENNA PIVOTED AT AFT END DFT YAVV STABILIZATION ±3° ANGLE CORRECTION N N NAV. SYS.





DESIGN EVALUATION

I. PHASE STABILITY MEASUREMENTS

A. FM DEVIATION OF ALL REFERENCE FREQUENCIES

B. JITTER OF ALL TIMING PULSES: < 4 NANOSEC

C. SYNTHETIC HOLOGRAMS GENERATED-RECORDED-CORRELATED

D. TRANSMITTER SIGNAL RADIATED TO IMILE REFLECTOR - RECORDED - CORRELATED

2. RANGE RESOLUTION

A. RECEIVER RESPONSE MEASURED

B. RESOLUTION CHECK OF RECEIVER WITH SYNTHETIC TARGETS-25 FT.-LINITED BY FIBER OPTICS

3. AZIMUTH RESOLUTION

A. SYNTHETIC HOLOGRAMS GENERATED. RECORDED. CORRELATED FOR FIOI CONDITIONS - SHARPENING RATIO OF 100/1 \approx 10 FT.

4. MISCELLANEOUS

A. INVESTIGATED PULSE OVERSHOOT - CORRECTED BY EXTENDING RECEIVER RESPONSE

B. INVESTICATED BEST RECORDER-FILM OPERATING POINT - 30% TRANSMISSION C. DYNAMIC RANGE OF RECEIVER-RECORDER-FILM COMBINATION > 20 db

D. AFFECT OF LIMITING TO INCREASE DYNAMIC RANGE;

JF - NO CLUTTER

VIDEO - BAD HARMONICS

UNGLASSIFIE

ENVIRONMENTAL TEST

NO SUSCEPTABILITY 1. RADIO INTERFERENCE (SYSTEM)

... MINOR RADIATION

2. EXPLOSION 3. VIBRATION

(SYSTEM)

RECEIVER
SYNCHRONIZER
NAV. TIE-IN
POWER SUPPLY
MODULATOR
DUPLEX. DRIVER
TWT
RECORDER

MTG BRACKET FAILED, CORRECTED. RECHECK NOISE FIGURE DETERIORATED. RECHECK MOVEMENT OF LENS, MIRROR

SPECIAL INVESTIGATION

10,20 ~ FROM POD 120,160~FROM ENGINE

> 4. CRASH SAFETY 5. TEMP. ALTITUDE

(SYSTEM) PLANNED (SYSTEM) PLANNED

UNGLASSIFIE

SQ. RIVER BRIDGE RE-INFORCED CONCRETE - 30 FT. WIDE RIVA BRIDGERE-INFORCED CONCRETE - 30 FT, WIDE STEPNEY'S LANE......NACADAM 30FT, CLEARING 45FT.

ANNAPOLIS - SOUTH RIVER LOC: . FLIGHT S-11

20,000 FT. ALT:

0.06 MEG WATTS PWR:

PULSE: 10.0 NANOSEC

ESTIM. EFFECTIVE RESOL.

A. LENS OPTICS RECORDER IN PLACE OF FIBER OPTICS
B. TRANSMITTER RESONANT PINC CONTRACTOR 2. MODIFICATIONS TO F-101 INSTALLATION O

TRANSMITTER RESONANT RING LENGTHENED - 10 TO 20 NANOSEC

AUTOMATIC STABILIZATION OF ANTENNA BY DFT

TARGETS: LEE AIRPORTGRASS STRIP, AMONG GRASS ANNAPOLIS - SOUTH RIVER 20,000 FT. : | | | ALT: 3. FLIGHT 5-33

O. II MEG WATTS PWR

PULSE: 20 NANOSEC

ESTIM. EFFECTIVE RESOL:

4. MODIFICATIONS TO F-101 INSTALLATION A. MOTION COMPENSATION SYSTEM

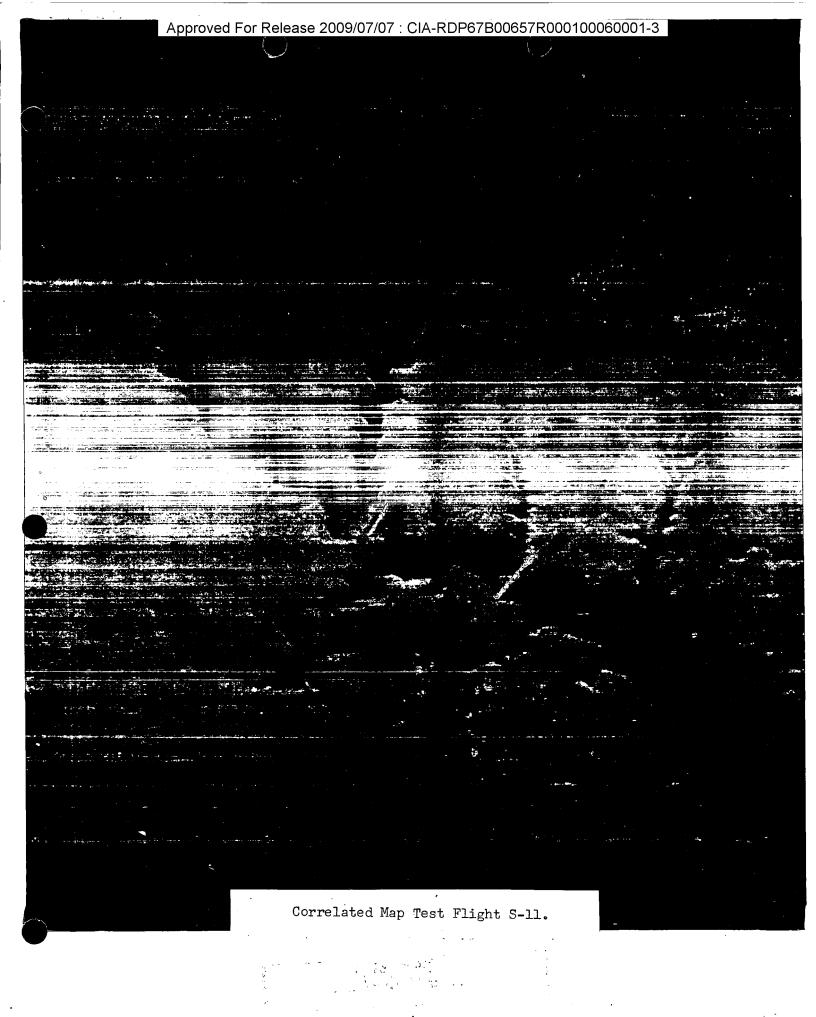
GROUND SPEED SIGNAL FOR RECORDER മ്

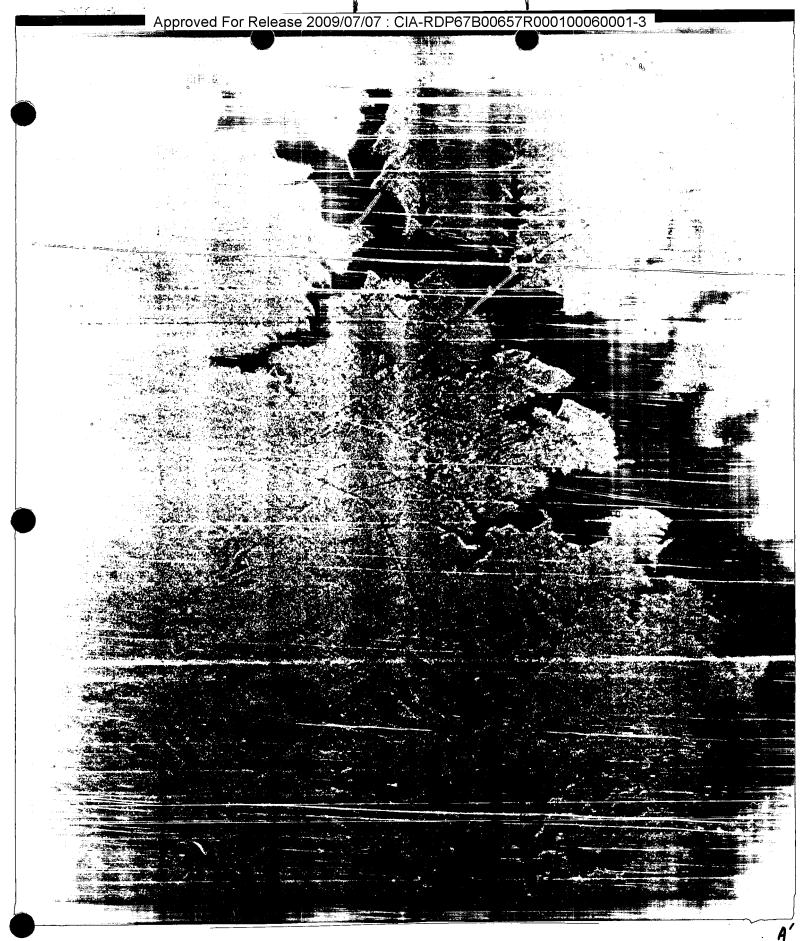
IF LIMITING

5. MEASURED RANGE FLIGHTS

CAMP CAMPBELL

WILCOX LAKE a a





Correlated Map Test Flight S-33.



Approved For Release 2009/07/07 : CIA-RDP67B00657R000100060001-3

Approved For Release 2009/07/07 : CIA-RDP67B00657R000100060001-3

TARGET DESCRIPTION

I- LEE AIRPORT STRIP

GRASS RUNWAY GRASS SURROUNDING

2- SOUTH RIVER BRIDGE

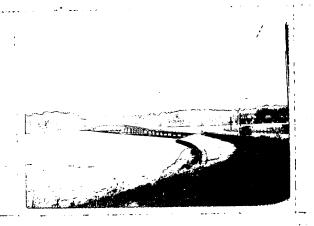
RE-INFORCED CONCRETE,

30 FT. WIDE

3 - RIVA BRIDGE







RE-INFORGED CONCRETE,

Approved For Release 2009/07/07 : CIA-RDP67B00657R000100060001-3

STEPNEYS LANE



MACADAM ROAD 30 FT. WIDE

CLEARING APPROX.45 FT. WIDE

BOAT PIERS









Correlated Data

South River Area	S-33 Prin	t or Film	S-11 Print	
1.airport runwaÿ	50 mils	(160')	not visible	
2.lower bridge	10 mils	(321)	50 mils	(1601.)
3. upper bridge	15 mils	(501)	50 mils	(1601)
4. macadam road	10 mils	(321)	off of picture	;
5.piers	7-10 mils	(221-321)	not visible	
minimum size point target	7 mils	(221)	15 mils	(501)
Annapolis Area				
academy object	15 mils	(501)	25 mils	(801)
Annapolis Bridge	7 mils	(221)	not visible	
Annapolis Bridge	10 mils	(321)	25 mils	(801)
boat dock objects	15 mils	(501)	25 mils	(801)
object on bay hook	10 mils	(321)	25 mils	(801)
minimum size point target	7 mils	(221)	15 mils	(501)

Scale Calculations

Scale 9" = 127 usec slant = 60,000' slant

Assume 1 mile dead slant range, so total slant range = 60,000 -6,000

9" = 57,000-1/2 (because only correlate 1/2 of data film)

$$1 \text{ mil} = 6.5 - 1/2 = 3.2$$

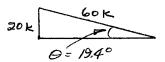
$$7 \text{ mils} = 45-1/2 = 22$$

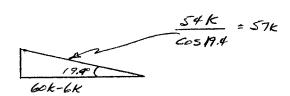
10 mils =
$$65-1/2 = 32$$

$$15 \text{ mils} = 100-1/2 = 50!$$

$$25 \text{ mils} = 160-1/2 = 80!$$

$$50 \text{ mils} = 320-1/2 = 160$$







ACCESSORY MATERIAL AND SERVICE

I FIELD SPARES

REVIEWING-CHANGING IN LINE WITH RADAR

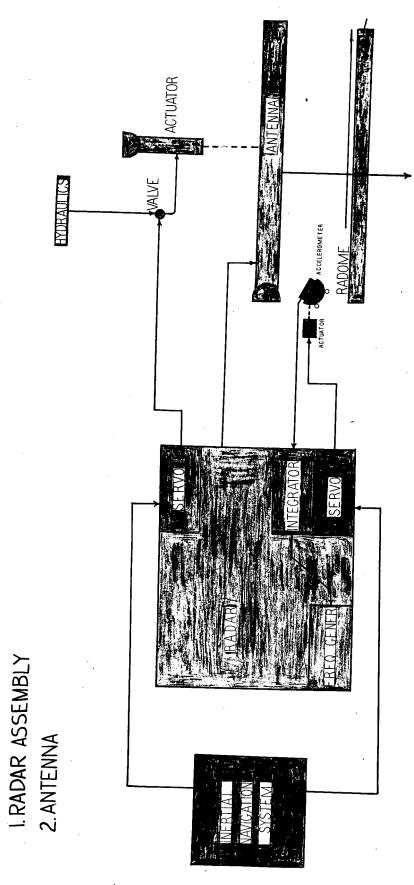
2. FIELD TEST EQUIPMENT

MODIFYING IN LINE WITH RADAR CHANGES

3. FIELD SERVICE

ENGINEERS WORKING WITH DESIGN GROUP





INSTALLATION

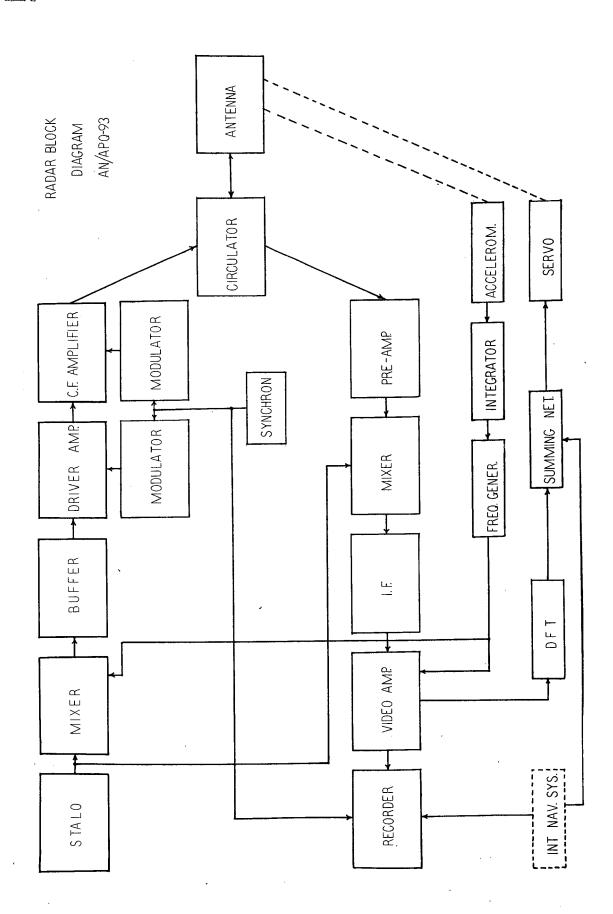
R.1803 RADAR, SURVEILL ANCE

R-1915 ANTENNA MOUNTING SPECIFICATION

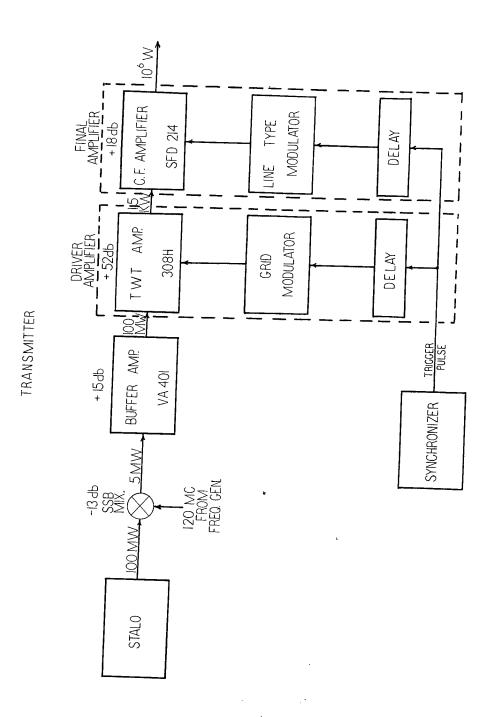
R-1916 ANTENNA ANGLE AND MOTION COMPENSATION

BLOCK DIAGRAM AN/APO-93

RADAR







UNCLASSIFIED

1/18	9400M 31.506 0.750es. 200es. 2500es.	-/406 -/506 -/506 -/506 /.3	2.0"/sec NOM. 1-10% 0.1% 250' x 9.5" 0.0005" 1.02My \$0x10-3ec 900 cps (Mox)
MANNE LEKS	ANTENNA FREQUENCY GAIN AZ BEAMWIDTH EL BEAMWIDTH EL PATTERN	HZ SIDELOBE EL SIDELOBE EL MAGE PATTERN VSWR RADOME LOSS (ONE WAY)	RECORDER FILM SPEED CONTROL FANSE CONTROL ACCURACY FILM CAPACITY CRT SPOT SIZE SWEED SPEED TRACK FREE.
NU VIOLE	240M 10 mats 30x10-3sc. 3327 118 watts	7.5016 2.1d6 (ave mar)	600 mc 47 mc 70 mc 400cps
TRANSMITTER	TN SINII ICK FREQUENCY PEAK POWER PULSE WIDTH PREPAGE POWER	CINE	MAYEGUIDE - 0.6506 (QUE MAY) STALO FREQUENCY 1-F 1-F 1-F 1-F 1NDEO AND BALDWIDTH 60 VIDEO AND BALDWIDTH 47 1MAGE FUTER BALDWIDTH 70 1MAGE REJECTION COND REF. OFFSET FREG. 400

SYSTEM NEIGHT

TRANSMITTER	* 0/2	SYSTEM FRAME
RECEIVER	27	FRAME TRUSS
TWT PREAMP	//	HUXILIARY RECORDER
RECORDER	/75*	TRESSURE SYSTEM
VIDEO AMPLIFIER	ŋ	707.42
SYNCHRONIZER	25	
NAV-TIE-12	*0×	
POWER SUPPLY	20/	
CONTROL PANEL	N	
ANTENNA	140 ×	

TOTAL WEIGHT - 119

683

* ESTIMATED WEIGHT

UNGLASSIFIE